



# Latest from Spin Research at Brunel

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## Spin testing the Super Emeraude



## Literature review revealed..

- No reliable risk reduction method exist
  Incremental, careful flight test approach
- Simplistic methods still being recommended for aircraft design, despite proven to be highly inaccurate
- Several cases of test pilots bailing out of prototypes due to unrecoverable spins

"Small changes might have a big impact"



# Spin Research Programme

- Camera tracking: capturing the spin motion

- Laser scan of aeroplane
- Video imagery from helicopter chase
- Camera tracking and visualization software
- The aerodynamic flow over wings and empennage
  - Tufts
  - Smoke
  - Several on-board cameras
  - Helicopter chase cameras

### Creating the CAD model

ĵ۲ Slingsby Firefly laser scanned in the hangar LN-TFF • 9 scan positions, 29.9 million points on aeroplane

## Video imagery from chase aircraft



#### Camera tracking





![](_page_7_Figure_0.jpeg)

#### Qualitative assessment of the flow

![](_page_8_Picture_1.jpeg)

- In-flight photographing of wool tufts using multiple cameras
- Areas of particular interest: airflow over wings and tail

![](_page_9_Picture_0.jpeg)

# Vortex Visualization – tufts 1/10 speed Slingsby Firefly

![](_page_10_Picture_1.jpeg)

![](_page_11_Figure_0.jpeg)

# Saab Safir

![](_page_12_Picture_1.jpeg)

#### Vortex Visualization – tufts 1/10 speed Saab Safir

![](_page_13_Picture_1.jpeg)

![](_page_14_Picture_0.jpeg)

![](_page_15_Picture_0.jpeg)

Image: GOES-13, NASA

# The breakthrough in understanding?

- The aeroplane in a spin must be considered as a rotating frame of reference
- The centrifugal, Coriolis and Euler accelerations do affect particles moving in a rotating system
- Hypothesis:
  - The turbulent layer on the upper surface, on the outside wing of a spinning aeroplane, is accelerated due to additional, spin induced accelerations

## Additional acceleration terms

- Coriolis acceleration varies with rotation rate and flow velocity:

$$-2\Omega \times v_{u}$$

- Centrifugal acceleration varies with rotation rate squared and radius:

$$-\Omega \times (\Omega \times r)$$

- Euler acceleration varies with time derivative of rotation rate and radius:

$$-\dot{\Omega} \times r$$

## Vortex Visualization using smoke

![](_page_18_Picture_1.jpeg)

![](_page_19_Picture_0.jpeg)

![](_page_20_Picture_0.jpeg)

![](_page_21_Picture_0.jpeg)

![](_page_22_Picture_0.jpeg)

![](_page_23_Picture_0.jpeg)

![](_page_23_Picture_1.jpeg)

#### 4 3,5 3 2,5 rad/sec 2 1,5 1 0,5 3rd turn 10 0 1st turn 2nd turn 4th turn 5th turn 6th turn 7th turn 2 20 0 4 6 8 12 14 16 18 Time (sec)

#### **Omega - Saab Safir, Right Hand Spins**

#### Flow over the horizontal tail

![](_page_25_Picture_1.jpeg)

## Flow over the horizontal tail cont.

- Spanwise flow on outside stabilizer
- RH spin Safir
- LH spin Slingsby

![](_page_26_Picture_4.jpeg)

![](_page_26_Picture_5.jpeg)

![](_page_27_Picture_0.jpeg)

![](_page_28_Figure_0.jpeg)

#### Slingsby 2 turn spin - Estimated pitching moment (Cm)

# Summary

- Flow visualization, using tufts and smoke, indicate the presence of 3 dimensional, complex flow fields
- Hypothesis: The turbulent layer on the upper surface, on the outside wing of a spinning aeroplane, is accelerated due to additional, spin induced accelerations
- These effects might be key to understanding the spin dynamics (e.g. turbulent flow impact on tail during spin, reversal of elevator effect and nose down pitching moment)

### Future research

We got observations and a hypothesis – we now need more data to validate the hypothesis

- More Instrumentation of research aeroplane
  - Flight data recorder
  - Air data probe
  - Differential pressure sensors
- Mathematical modelling

![](_page_31_Picture_0.jpeg)

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